

ELASTIC PLATE THEORY APPLIED TO PCC PAVEMENTS

I Summary of elastic plate theory

Elastic plates in bending can be characterized as:

- 1) Thick Plates for which transverse shear is considered;
- 2) Thin plates for which bending and membrane stresses are considered;
- 3) Medium thick plates for which only bending is considered; (thickness between 1/20 to 1/100 of span length)

If it is assumed that transverse deflections are small so that in-plane forces produced by stretching of the middle plane can be ignored, then the medium thick plate theory is applicable.

Three basic assumptions:

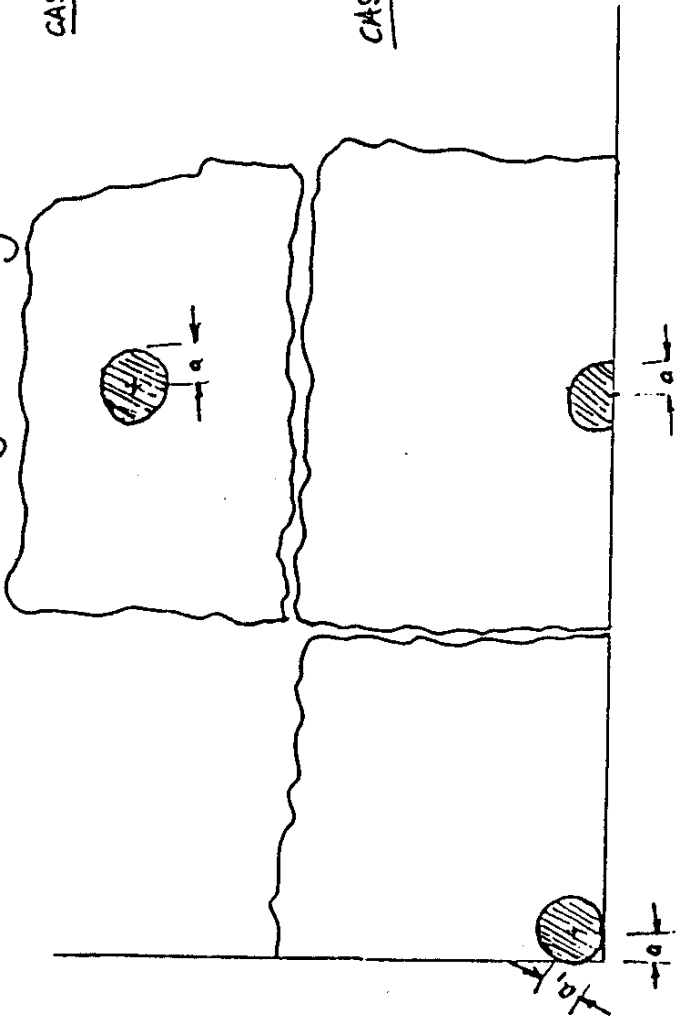
1. Planes normal to the middle plane of the plate before bending remain normal after bending (this is equivalent to an assumption of no shear deformation, $\gamma_{xz} = \gamma_{yz} = 0$);
2. Forces normal to the plate surfaces are small when compared with bending stresses and can be ignored ($\sigma_z = 0$);
3. Stresses normal to a plane normal to the middle plane of the plate are small and can be ignored ($\Sigma N_x = \Sigma N_y = 0$).

All stresses are defined per unit of length of the plate mid-surface.

①

• Westergaard (1926) :

- Three cases of loading Investigated



CASE 1 : INTERIOR : critical tension occurs at bottom of slab under center of circle

CASE 2 : EDGE : critical tension occurs at bottom of slab under center of semi-circle.

CASE 3 : CORNER : critical tension occurs at the top of slab. The resultant of applied pressure is at center of circle ; $a_1 = a\sqrt{2}$

TABLE 1

WESTERGAARD EQUATIONS FOR INTERIOR LOADING

(a) Maximum Bending Stress, σ_1

ORDINARY THEORY:	$BS10T = \frac{3P(1+\mu)}{2nh^2} [ln(2\ell/a) + 0.5 - \gamma] + BS120T$	(1-1) Theoretically rigorous: USE!
SPECIAL THEORY:	$BS1ST = \frac{3P(1+\mu)}{2nh^2} [ln(2\ell/h) + 0.5 - \gamma] + BS12ST$	(1-2) Crude adjustment for neglected shear stresses: AVOID!
FOR SQUARE:	$BS1SQ = \frac{3P(1+\mu)}{2nh^2} [ln(2\ell/c') + 0.5 - \gamma] + BS12SQ$	(1-3) For comparison w/ FEM, etc.

SUPPLEMENTARY, σ_2 (Ordinary Theory)	$BS120T = \frac{3P(1+\mu)}{64h^2} [(n/\ell)^2 - 1 = \frac{3P(1+\mu)}{2\pi h^2} \left\{ \frac{\pi}{32} \left(\frac{a}{h} \right)^4 \right\}]$	(1-4)
SUPPLEMENTARY, σ_2 (Special Theory)	$BS12ST = \frac{3P(1+\mu)}{64h^2} [(n/\ell)^2 - 1]$	(1-5) Additional terms to truncated infinite series (usually negligible)
SUPPLEMENTARY, σ_2 (For Square)	$BS12SQ = \frac{3P(1+\mu)}{64h^2} [(c'/\ell)^2 - 1]$	(1-6)

(b) Maximum Deflection, δ_1

CIRCLE:
$$DEFIC = \frac{P}{8k\ell^2} [1 + (1/2\pi) \{ ln(n/2\ell) + \gamma - 5/4 \} (n/\ell)^2] \quad (1-7)$$

- P : total applied load;
- E : slab Young's modulus;
- μ : slab Poisson's ratio;
- h : slab thickness;
- k : modulus of subgrade reaction;
- a : radius of circular load; $b = \sqrt{a^2 + h^2}$ $-0.675h$, if $a < 1.724h$
- c : side length of square load; a $-0.675h$, if $a > 1.724h$
- $\ell^* = \frac{Eh^3}{12(1-\mu^2)k}$ $c' = \frac{c^2/4 - 1}{\sqrt{2}}$
- γ : Euler's constant (≈ -0.57721566490)

Ref: "Westergaard Solutions Reconsidered," by A.M. Ioannides, M.R. Thompson and E.J. Barenberg (1985)

TABLE 2
WESTERGAARD EQUATIONS FOR EDGE LOADING

(a) Maximum Bending Stress, σ_e

ORDINARY THEORY: (Semi-circle) BSEWOT = $0.529 (1 + 0.54u) \frac{P}{h^2} [\log_{10} \left\{ \frac{Eh^3}{ka_2^3} \right\} - 0.71]$ (E-1)

SPECIAL THEORY: (Semi-circle) BSEWST = $0.529 (1 + 0.54u) \frac{P}{h^2} [\log_{10} \left\{ \frac{Eh^3}{kb_2^3} \right\} - 0.71]$ (E-2)

"NEW" FORMULA: (Circle) BSEIC = $\frac{3(1+u)P}{\pi(3+u)h^2} \left[\ln \frac{Eh^3}{100ka_2^3} + 1.84 - 4u/3 + \frac{(1-u)}{2} + 1.18(1+2u)(a/\ell) \right]$ (E-3)

"NEW" FORMULA: (Semi-circle) BSEIS = $\frac{3(1+u)P}{\pi(3+u)h^2} \left[\ln \frac{Eh^3}{100ka_2^3} + 3.94 - 4u/3 + 0.5(1+2u)(a_2/\ell) \right]$ (E-4)

SIMPLIFIED "NEW" FORMULA: (Semi-circle) BSELS = $\frac{-6P}{h^2} (1+0.5u) [0.489 \log_{10} (a_2/\ell) - 0.091 - 0.027(a_2/\ell)]$ (E-5)

SIMPLIFIED "NEW" FORMULA: (Circle) BSELC = $\frac{-6P}{h^2} (1+0.5u) [0.489 \log_{10} (a/\ell) - 0.012 - 0.063(a/\ell)]$ (E-6)

(b) Maximum Deflection, δ_e

ORIGINAL FORMULA: DEF EW = $\frac{1}{\sqrt{6}} (1 + 0.4u) \frac{P}{k\ell^2}$ (E-7)
ignores load distribution

"NEW" FORMULA: (Circle) DEFEIC = $\frac{P\sqrt{2+1.2u}}{\sqrt{Eh^3k}} [1 - (0.76 + 0.4u)(a/\ell)]$ (E-8)

"NEW" FORMULA: (Semi-circle) DEFEIS = $\frac{P\sqrt{2+1.2u}}{\sqrt{Eh^3k}} [1 - (0.323 + 0.17u)(a_2/\ell)]$ (E-9)

SIMPLIFIED "NEW" FORMULA: (Semi-circle) DEFELS = $\frac{1}{\sqrt{6}} (1 + 0.4u) \frac{P}{k\ell^2} [1 - 0.323(1+0.5u)(a_2/\ell)]$ (E-10)

SIMPLIFIED "NEW" FORMULA: (Circle) DEFELC = $\frac{1}{\sqrt{6}} (1 + 0.4u) \frac{P}{k\ell^2} [1 - 0.760(1+0.5u)(a/\ell)]$ (E-11)

a_2 : radius of semi-circle;

$b_2 = \sqrt{1.6 a_2^2 + h^2} - 0.675h$, if $a_2 < 1.724h$

= a_2 if $a_2 > 1.724h$

See Table 1 for other symbols

... functions: PROBABLY

Losberg, 1960
 Westerg. 1948

only about 2% different from full expressions

TABLE 5

EQUATIONS PROPOSED FOR THE CORNER LOADING CONDITION

Deflection:

$$\delta_c = \frac{P}{kl^2} \left(1.1 - 0.88 \frac{a_1}{l} \right) \quad \text{Westergaard [5]} \quad (C-1)$$

Stress:

$$\sigma_c = \frac{3P}{h^2} \quad \text{Early simplistic analysis} \quad \text{Goldbeck [25] ; Older [26]} \quad (C-2)$$

$$\sigma_c = \frac{3P}{h^2} \left[1 - \left(\frac{a_1}{l} \right)^{0.6} \right] \quad \text{Westergaard [5]} \quad (C-3)$$

Crude numerical analysis

$$\sigma_c = \frac{3P}{h^2} \left[1 - \left(\frac{a_1}{l} \right)^{0.6} \right] \quad \text{Bradbury [15]} \quad (C-4)$$

$$\sigma_c = \frac{3P}{h^2} \left[1 - \left(\frac{a_1}{l} \right)^{1.2} \right] \quad \text{Kelley [16] , Teller and Sutherland [13]} \quad (C-5)$$

$$\sigma_c = \frac{3.2P}{h^2} \left[1 - \left(\frac{a_1}{l} \right) \right] \quad \text{Spanzler [29]} \quad (C-6)$$

$$\sigma_c = \frac{4.2P}{h^2} \left[1 - \frac{\sqrt{(a/2)}}{0.925 + 0.22 \left(\frac{a}{l} \right)} \right] \quad \text{Pickett [23]} \quad (C-7)$$

Empirical adjustments to Westergaard's equation (C-3)

Distance to point of max. stress along corner angle bisector:

$$X_1 = 2\sqrt{(a_1 l)} \quad \text{Westergaard [5]} \quad (C-8)$$

where a : radius of circular load tangent to both edges at corner
 a_1 : distance to point of action of resultant along corner angle bisector
 $= \frac{\sqrt{2}}{2} a$

(See Table 1 for other symbols.)

Ioannides (1984) [based on FEM]:

$$\text{USE! } \left\{ \begin{array}{l} \delta_c = \frac{P}{kl^2} \left\{ 1.205 - 0.69 \left(\frac{c}{l} \right) \right\} = \frac{P}{kl^2} \left[1.2 - 0.88 \left(\frac{a_1}{l} \right) \right] \\ \sigma_c = \frac{3P}{h^2} \left\{ 1.0 - \left(\frac{c}{l} \right)^{0.72} \right\} = \frac{3P}{h^2} \left[1 - \left(\frac{a_1}{l} \right)^{0.64} \right] \\ X_1 = 1.80 c^{0.32} l^{0.59} \end{array} \right. \quad c: \text{ side length of square load}$$

Radii of Relative Stiffness						
Thickness inches	Concrete Modulus = 4 million psi, Poisson's ratio = .15					
	Subgrade Support (k)					
	50 psi/in	100 psi/in	200 psi/in	300 psi/in	400 psi/in	500 psi/in
4.0	25.70	21.61	18.18	16.42	15.28	14.45
4.5	28.08	23.61	19.85	17.94	16.69	15.79
5.0	30.39	25.55	21.49	19.42	18.07	17.09
5.5	32.64	27.44	23.08	20.85	19.41	18.35
6.0	34.84	29.30	24.63	22.26	20.72	19.59
6.5	36.99	31.11	26.16	23.64	22.00	20.80
7.0	39.11	32.89	27.65	24.99	23.25	21.99
7.5	41.19	34.63	29.12	26.32	24.49	23.16
8.0	43.23	36.35	30.57	27.62	25.70	24.31
8.5	45.24	38.04	31.99	28.91	26.90	25.44
9.0	47.22	39.71	33.39	30.17	28.08	26.55
9.5	49.17	41.35	34.77	31.42	29.24	27.65
10.0	51.10	42.97	36.14	32.65	30.39	28.74
10.5	53.01	44.57	37.48	33.87	31.52	29.81
11.0	54.89	46.16	38.81	35.07	32.64	30.87
11.5	56.75	47.72	40.13	36.26	33.74	31.91
12.0	58.59	49.27	41.43	37.44	34.84	32.95
12.5	60.41	50.80	42.72	38.60	35.92	33.97
13.0	62.22	52.32	43.99	39.75	36.99	34.99
13.5	64.00	53.82	45.26	40.89	38.06	35.99
14.0	65.77	55.31	46.51	42.02	39.11	36.99
14.5	67.53	56.78	47.75	43.15	40.15	37.97
15.0	69.27	58.25	48.98	44.26	41.19	38.95
15.5	70.99	59.70	50.20	45.36	42.21	39.92
16.0	72.70	61.13	51.41	46.45	43.23	40.88
16.5	74.40	62.56	52.61	47.54	44.24	41.84
17.0	76.08	63.98	53.80	48.61	45.24	42.78
17.5	77.75	65.38	54.98	49.68	46.23	43.72
18.0	79.41	66.78	56.15	50.74	47.22	44.66
18.5	81.06	68.17	57.32	51.80	48.20	45.59
19.0	82.70	69.54	58.48	52.84	49.17	46.51
19.5	84.33	70.91	59.63	53.88	50.14	47.42
20.0	85.94	72.27	60.77	54.91	51.10	48.33

Radius of Relative Stiffness = l

$$l = [E \cdot h^3 / (12 \cdot (1 - \mu^2))]^{1/4}$$

E v k

According to Boussinesq; for $\mu = .5$

$$w = \frac{1.5p \cdot a}{E}$$

where

p = unit pressure over circular area

a = radius of circular loaded area

E = modulus of the elastic mass

By definition; $k = \frac{p}{w}$ (Westergaard)

Rearranging the Boussinesq Equation

$$\frac{p}{w} = k = \frac{E}{1.5 \cdot a}$$

Therefore, either E or k must be
a function of the loaded area "a"

E v k Continued

From Slab Theory:

$$w_h = \frac{P \cdot l_k^2}{8 \cdot D} \quad \text{for slab on dense liquid}$$

$$w_e = \frac{P \cdot l_e^2}{3\sqrt{3} \cdot D} \quad \text{for slab on elastic solid}$$

but $w_h = w_e$

Therefore:

$$\frac{P \cdot l_k^2}{8 \cdot D} = \frac{P \cdot l_e^2}{3\sqrt{3} \cdot D}$$

or

$$\frac{l_k^2}{8} = \frac{l_e^2}{3\sqrt{3}} \quad \Rightarrow \quad .6495 l_k^2 = l_e^2$$

$$l_k = \left[\frac{D}{k} \right]^{1/4}$$

$$l_e = \left[\frac{2 \cdot D}{E_s / (1 - \mu_s^2)} \right]^{1/3}$$

E v k Continued

$$0.6495 \cdot I_k^2 = I_e^2$$

Where:

$$I_k = \left[\frac{E \cdot h^3}{12 \cdot (1 - \mu_c^2) \cdot k} \right]^{1/4} \quad \text{and} \quad I_e = \left[\frac{2 \cdot E \cdot h^3 \cdot (1 - \mu_s^2)}{12 \cdot (1 - \mu_c^2) \cdot E_s} \right]^{1/3}$$

Substituting

Poisson's Ratio of Soil = 0.5

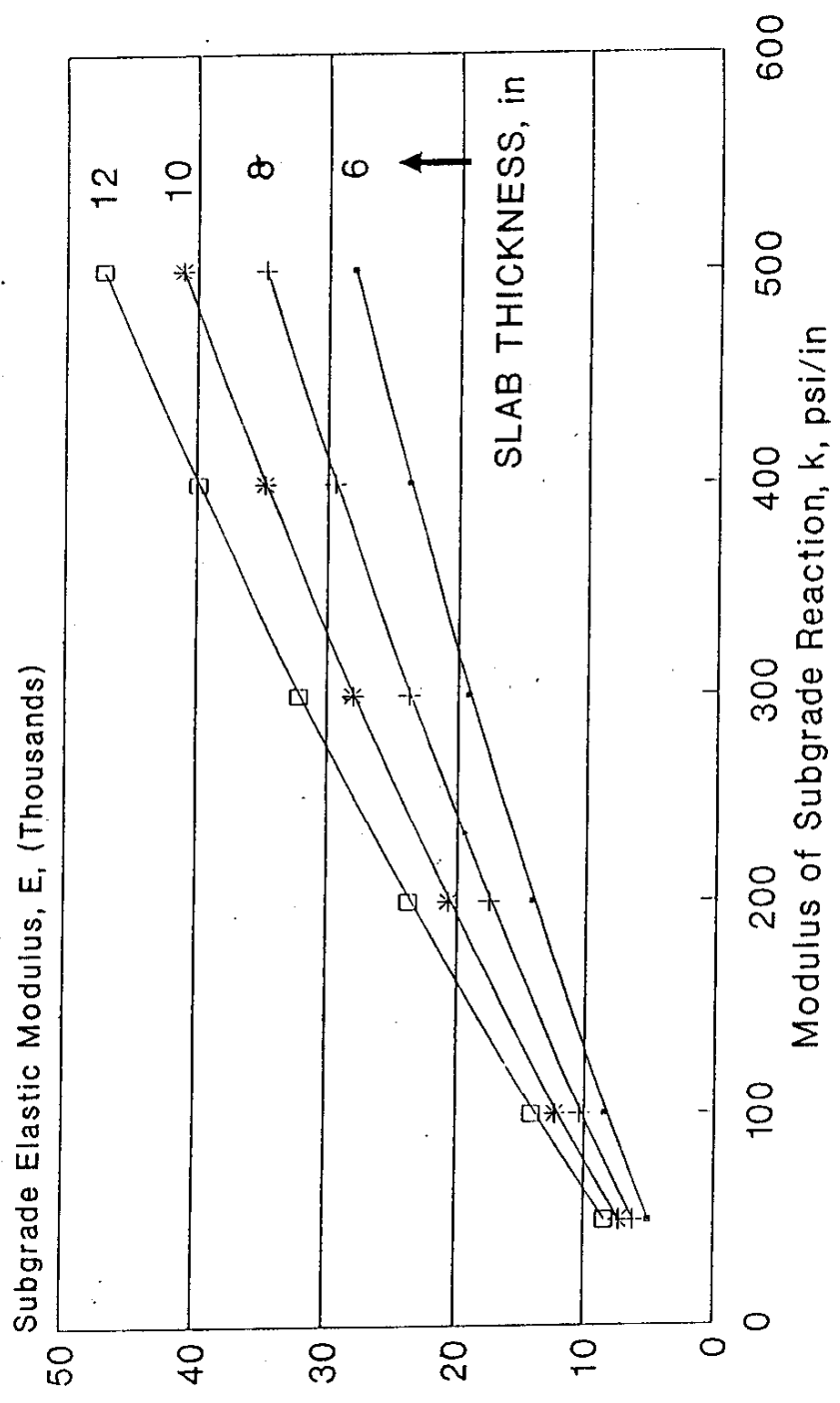
Poisson's Ratio of Concrete = 0.15

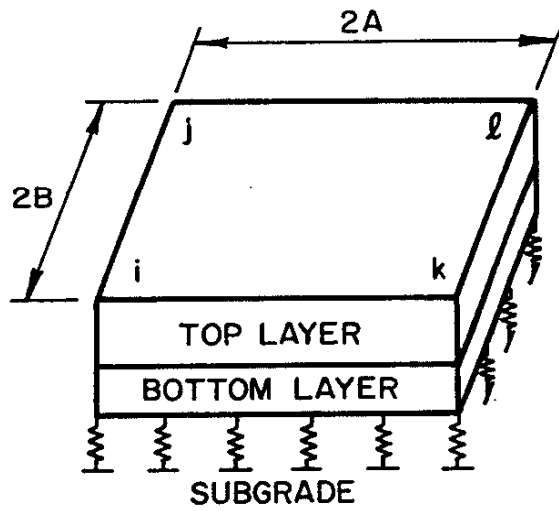
Modulus of Concrete = 4,000,000 psi

and Clearing the Equation Yields

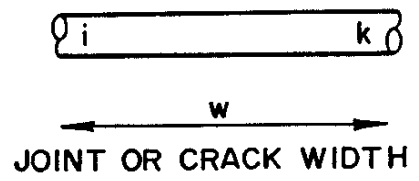
$$E_s^{4/3} = 283.7 \cdot h \cdot k$$

E soil versus k soil For Various Slab Thicknesses

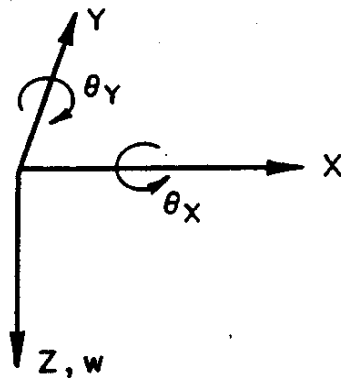




a) PLATE ELEMENT



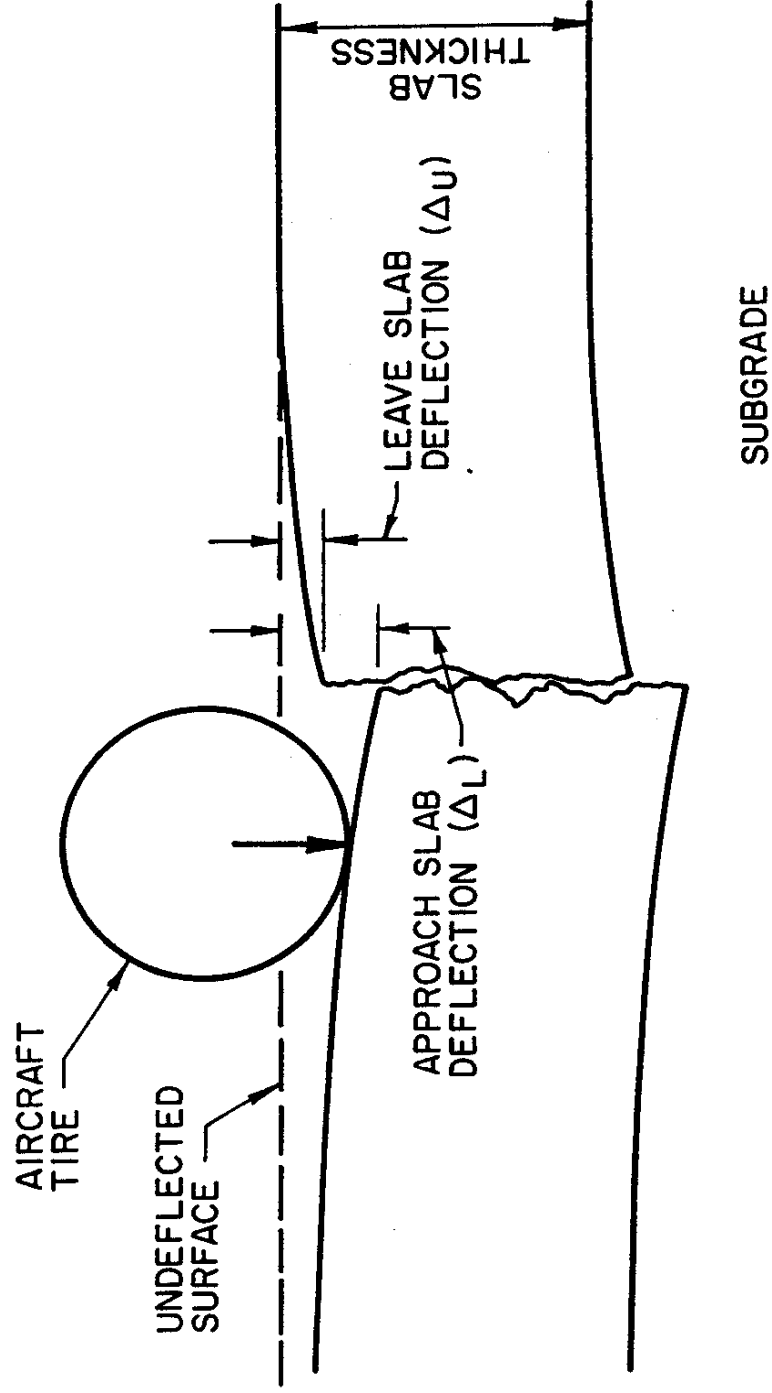
b) BAR ELEMENT



c) SPRING ELEMENT

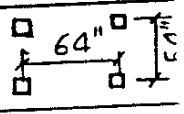
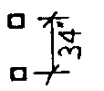
FINITE-ELEMENT MODEL OF PAVEMENT SYSTEM

$$\text{LOAD TRANSFER EFFICIENCY (\%)} = \frac{\Delta U}{\Delta L} \times 100$$



THE CONCEPT OF JOINT LOAD TRANSFER EFFICIENCY

MAXIMUM EDGE STRESS vs. JOINT EFFICIENCY

A/C	LOAD PER TIRE	TIRE PRESSURE	WHEEL CONFIGURATION
DC-10	52.6 ^K	172 PSI	
B-727	40.0 ^K	170 PSI	
B-747	45.5 ^K	207 PSI	